

Advanced compressive shear testing and in-situ failure analysis of glass-laminates

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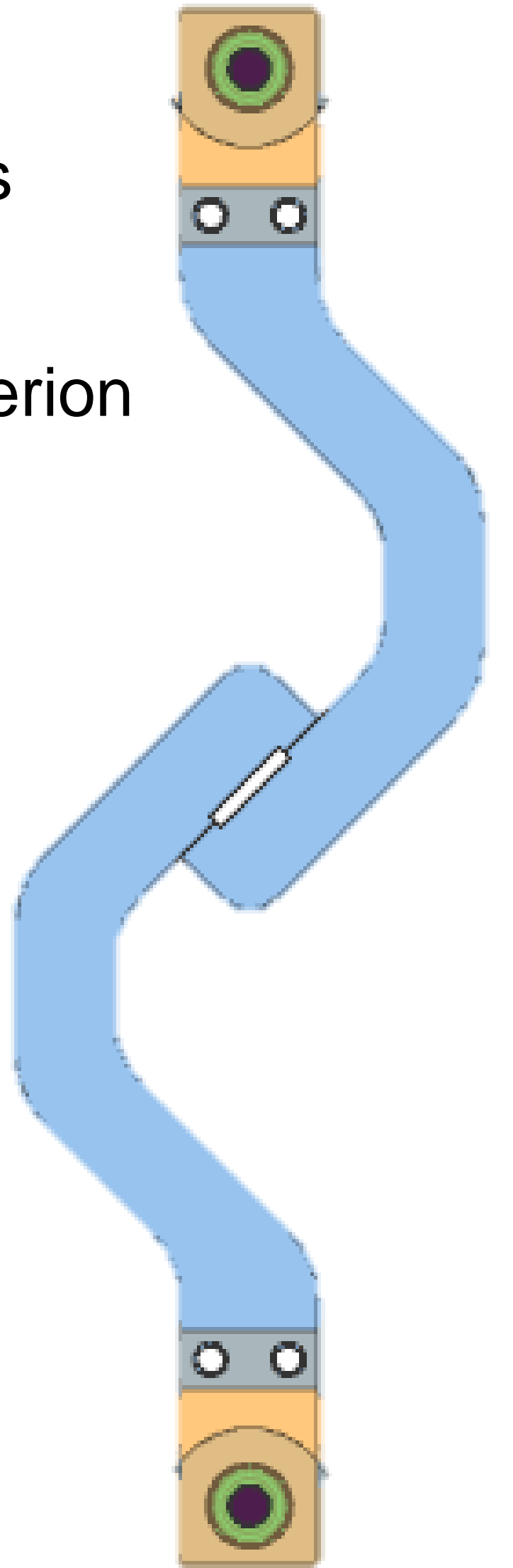
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Introduction

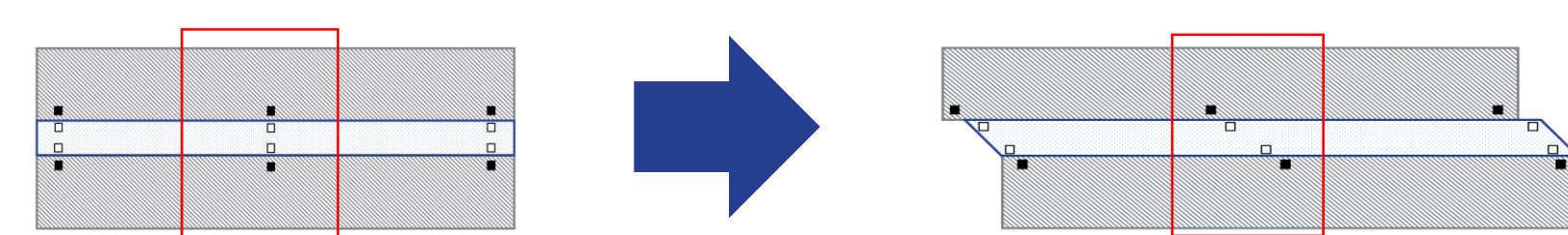
- Compressive shear testing (CST) was applied to characterize the delamination performance of glass-encapsulant laminates
- Digital image correlation (DIC) Aramis (GOM) was used to identify the point of delamination
- Point tracking of DIC generated image series was combined with a finite element model (FEM) to define a delamination criterion

Experimental and Modelling

- Glass laminates were manufactured using an UV-transparent, peroxide crosslinking encapsulant (based on ethylene vinyl acetate copolymer (EVA)) at 155°C (curing time: 15 min, pressure: 800 mbar); specimens of 25x25 mm² were obtained using water jet cutting
- CST was performed in tensile mode on a Zwick-Roell Z020 equipped with Aramis at a loading rate of 1 mm/min
- GOM Correlate was used to analyse the image series generated by Aramis:
 - Point tracking was performed on 12 (initial) vertically aligned surface points
 - Three regions (left, middle, right) consisting of four points (two on glass and two on EVA) were assessed
 - Output: vertical and horizontal displacement of each point with corresponding shear angles
- Two approaches were developed to evaluate the point tracking data; a FEM was built in order to estimate the delamination load (F_{delam})

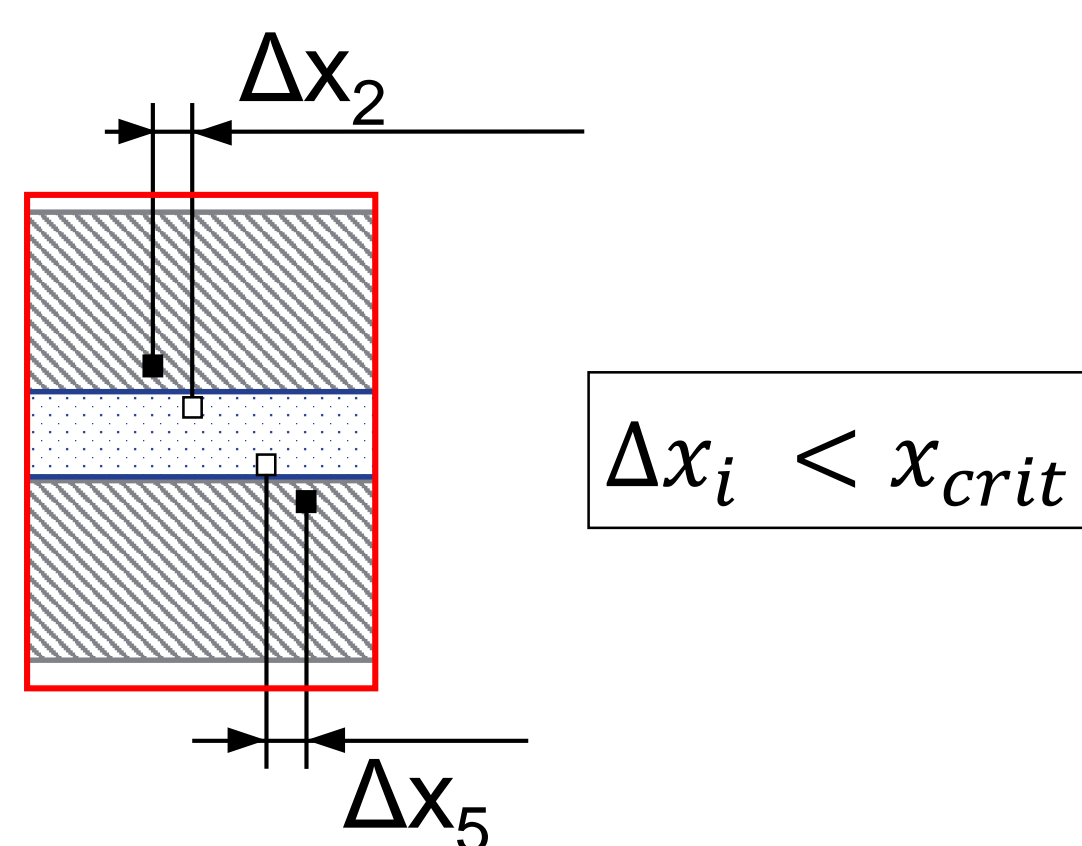


Evaluation Approaches



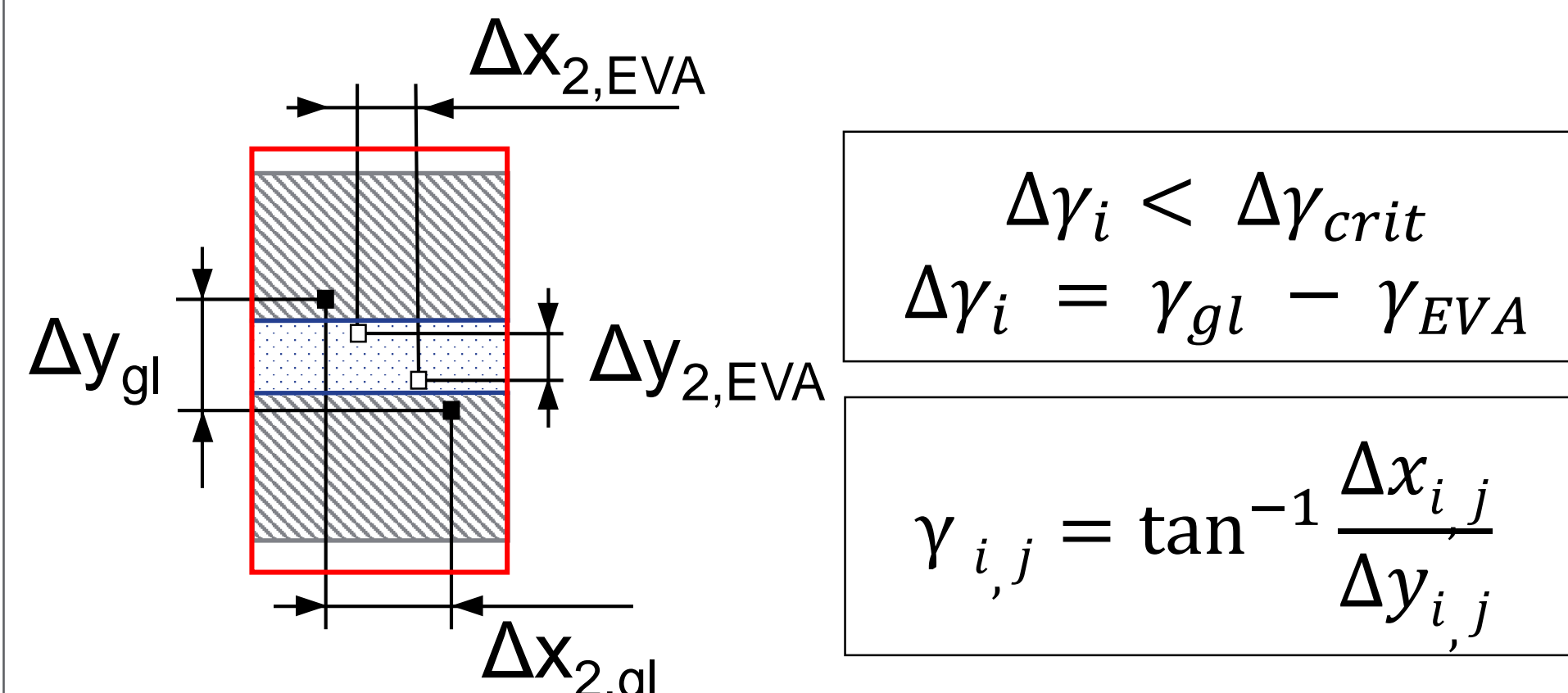
Approach 1

Consideration of the change in x



Approach 2

Comparison of the shear angle in EVA (γ_{EVA}) to the shear angle between the glass substrates (γ_{gl})



FEM

Pure shear of a CST specimen

Linear elastic material models for EVA and glass
EVA: $E^*=10$ MPa, $\eta = 0.45$, $\rho = 0.92$ g/cm³ (*obtained from Wallner et al., 2010)

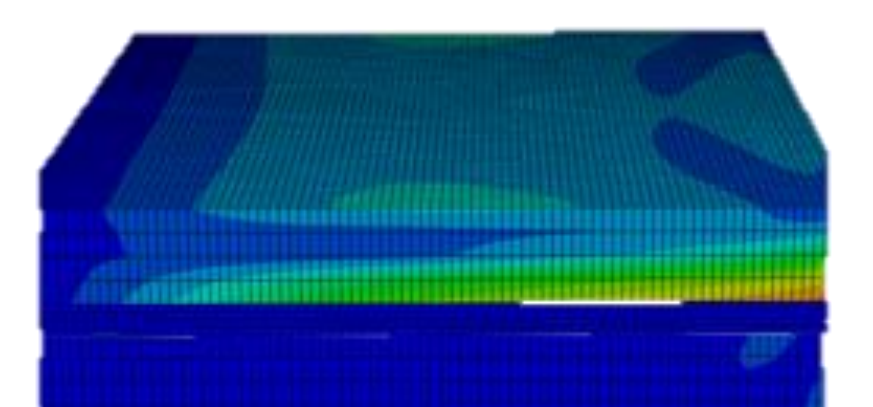
Glass: $E=70$ GPa, $\eta = 0.23$, $\rho = 2.5$ g/cm³

Two sets of fracture mechanical (FM) properties (adopted from He et al., 2018) were applied:

- Set 1: $G_{Ic,1} = 99$ J/m²; $T_{I,1} = 0.2$ MPa
- Set 2: $G_{Ic,2} = 1200$ J/m²; $T_{I,2} = 4$ MPa

$$G_{IIc} = G_{IIIc} = 3 * G_{Ic}$$

$$T_{II} = T_{III} = 3 * T_I$$

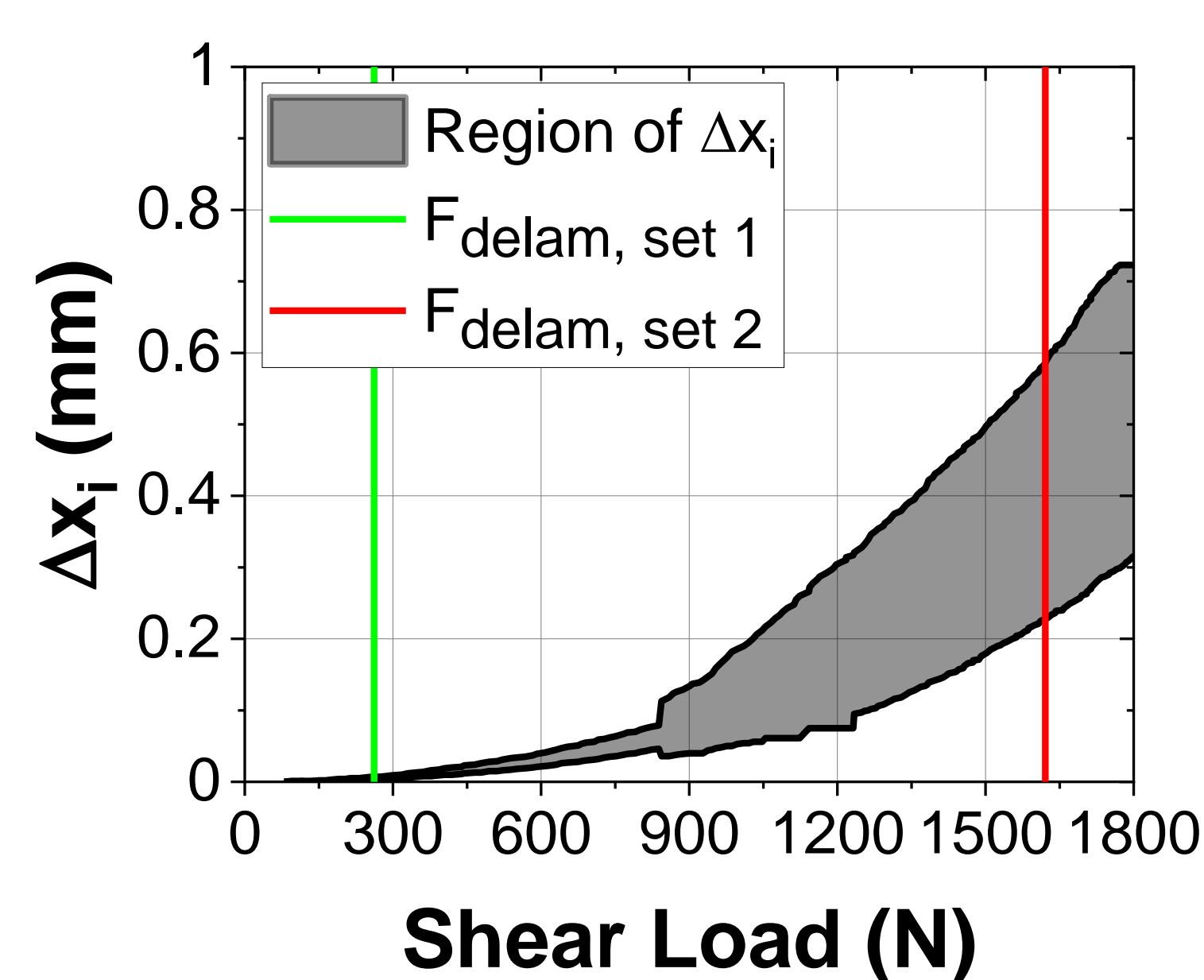


Results & Discussion

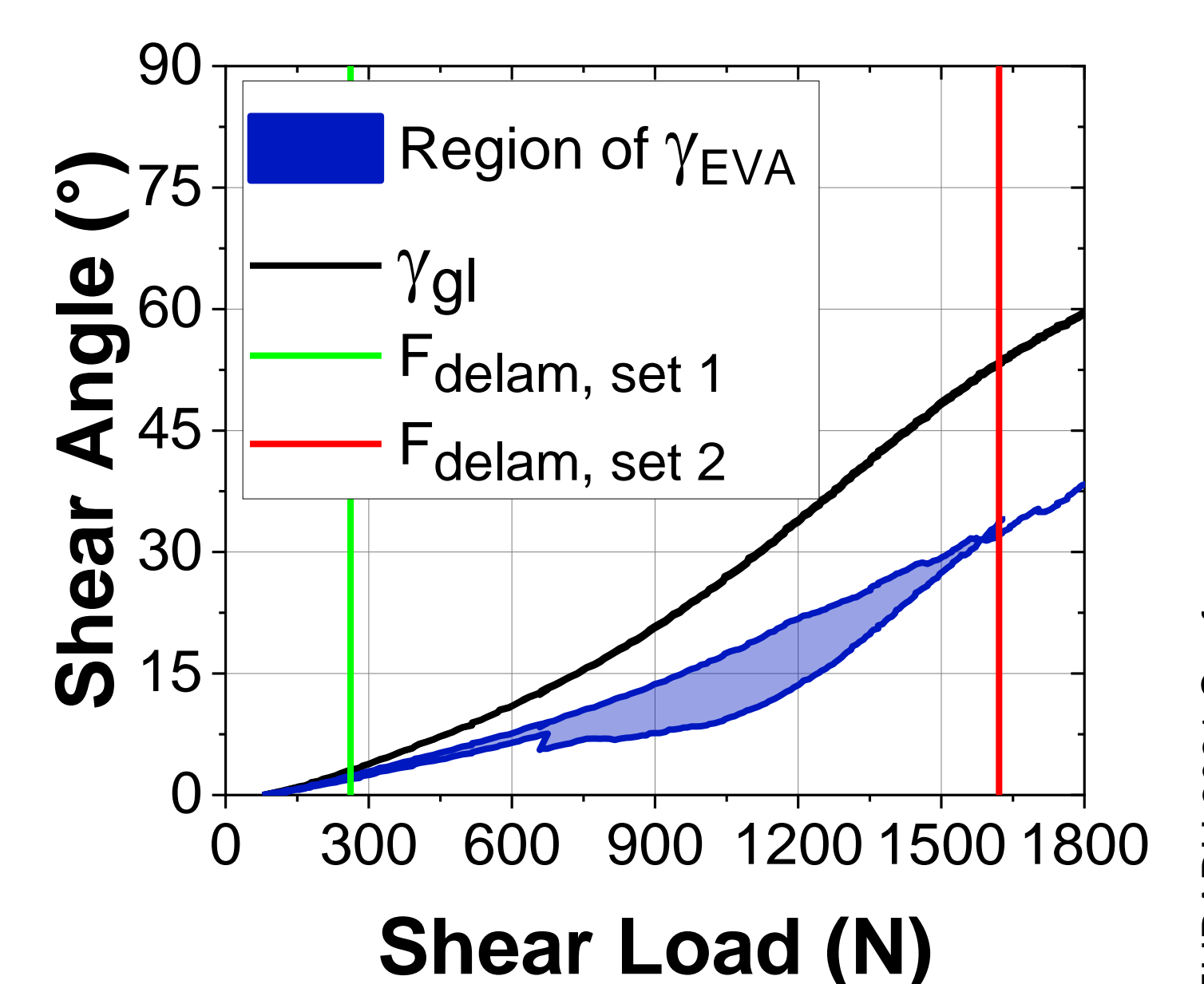
- Point of delamination did not correspond to maximum load
- Complex failure (including glass fracture) of CST specimens
- No delamination criteria could be defined with approach 1 or 2
- Based on the failure loads (262 and 1621 N) deduced by FEM, four delamination criteria were obtained

FEM-Input	FEM-Result	Delamination at	
	F_{delam} (N)	x_{crit} (mm)	$\Delta\gamma_{crit}$ (°)
Set 1	262	0.01	1.41
Set 2	1621	0.36	19.78

Approach 1



Approach 2



Outlook

- Future work will focus on width tapered cantilever beam (WTCB) tests to generate accurate fracture mechanical parameters of the laminates and a study of the effect of damp heat ageing on shear performance

Acknowledgement

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